## COMPARATIVE PERFORMANCE OF VARIOUS CONTRACTION PROFILES

S.Ramaseshan CRE(Aircraft) CEMILAC Bangalore India

M.A.Ramaswamy
Aerospace department
Indian Institute of Science
Bangalore
India

#### **ABSTRACT**

This paper brings out the studies carried out on the performance of various contraction profiles. A new family of profiles "Curvature Defined Profiles", has shown better performance than the earlier profiles considered. A systematic comparison of the performance of 2-D contraction profiles has been made. The new family of profiles offer very low exit flow non-uniformity compared to all the other profiles for the same margin of separation.

#### INTRODUCTION

It is well known that all the finite length contractions give raise to some amount of adverse pressure gradient on the wall near the entry and exit sections and also non-uniformity of flow at the exit station and these features cannot be avoided. Therefore the objective of practical contraction shapes is to keep the adverse pressure gradients small enough to avoid boundarylayer separation and minimise the non-uniformity of flow at the exit for a given length and contraction ratio. Bradshaw(Ref 2) has mentioned that atleast first and second derivatives of the wall profile should be continuous and they be zero at the entry and exit stations. However the practical profiles which have been proposed in the past by Morel(Ref 3) and Downie(Ref 1) have discontinuity in the second derivatives of the wall profiles.

It was felt, that though the boundary layer might somewhat effectively smoothen out the discontinuity of the second derivative of the profile, it was still desirable to avoid such discontinuities. Strictly speaking, boundary layer equations may not be valid if large jumps in curvature occurs.

It was therefore felt that the following families of smooth profiles need to be looked into and their performance compared with the cubic and elliptic profiles for the 2-D case. One such profile earlier proposed is a profile with fifth degree polynomial which is unique for a given contraction ratio and length. One of another new profiles we considered is the matched sine profiles which has the location of the matching station as a free parameter. Another family of new profiles, say, "Curvature Defined Profiles" which has the flexibility, viz, the matching location, the ordinate and slope at the matching point as in the elliptical profiles have been analysed.

For obtaining the performance of these profiles a computational method which numerically solves the Laplace equation by the line relaxation method was developed. It is shown that "Curvature Defined profiles" are considerably superior to all the other profiles considered here.

# CHARACTERISTICS OF VARIOUS PROFILES Fifth order profile

y=1.0-
$$(7.5/L^3)x^3 + (11.25/L^4)x^4 - (4.5/L^5)x^5$$
 for  $0 \le x \le L$   
Cubic profile

$$y=1.0-(1-r/L)x^3/X_m^2$$
 for  $0 \le x \le X_m$ ;  $y=r+(1-r/L)(L-x)^3/(L-X_m)^2$  for  $X_m \le x \le L$  Sine profile

$$\begin{array}{lll} y = 1.0 - (1 - r/L)x + (1 - r/L)(X_m/\pi) \sin \pi/x & \text{for } 0 \leqslant x \leqslant X_m \\ 1.0 - (1 - r/L)x - (1 - r/L)(L - X_m/\pi) & \sin \pi(x - X_m)/(L - X_m) & \text{for } X_m \leqslant x \leqslant L \end{array}$$

# Curvature defind profiles

$$\begin{array}{lll} y'' = -k_1 \ x^{n1} & \text{for} \ 0 \leqslant x \leqslant x_1; & y'' = -k_1 x_1^{n1} (1 - (x - x_1) / (X_m - x_1)) & \text{for} \ x_1 \leqslant x \leqslant X_m \\ & = \ k_2 (L - x_2)^{n2} ((x - X_m) / (x_2 - X_m)) & \text{for} \ X_m \leqslant x \leqslant x_2; & y'' = \ k_2 (L - x)^{n2} & \text{for} \ x_2 \leqslant x \leqslant L \end{array}$$

where, k<sub>1</sub>, k<sub>2</sub>, x<sub>1</sub>,x<sub>2</sub> are functions of X<sub>m</sub>,H,T,r and L for a given n1 and n2.

For all the above profiles X<sub>m</sub>=matching location; H=Height at the matching location; T=slope at the matching location; L=length of the contraction profile; c=contraction ratio; r=exit co-ordinate=1/c

## PERFORMANCE OF VARIOUS PROFILES

Laplace equation in stream function is solved in the transformed rectangular computational plane. The code was validated by comparing the results with the exact solution for a 2-D contraction duct with straight wall contours(Fig 1). The performance of the contraction profile is assessed by the non-uniformity of velocity at the exit(U2) as defined by Morel(Ref3) and the avoidance of boundarylayer separation based on Stratford's turbulent separation criterion (Ref

4) which is given below. Here the results are discussed for a contraction ratio of 4 only.  $C_{p}(s \, dC_{p} / ds)^{0.5} < 0.35 \, (10^{-6} \, Re \, s)^{0.1} \qquad (1)$ The length along the wall 's' is non-dimensionalised with respect to entry ordinate and Re is based entry ordinate. In the analysis Re is taken as 10<sup>6</sup> and (RHS-LHS) of equation (1) is defined as the margin of separation, S<sub>m</sub>(negative values mean separation present) Re is Reynolds Number, Cp is pressure coefficient and dCp/ds is the pressure gradient.

#### Fifth order profile

This has a unique profile and gives  $U_2 = 1.4\%$  and separation margin  $S_m = 0.05$ 

### Cubic and Sine profiles

For these profiles only one parameter, the matching location X<sub>m</sub> can be varied. It was found that as X<sub>m</sub> increases S<sub>m</sub> improves but non-uniformity also increases. The performance of cubic and sine profiles are shown in Fig.3 and they are very close to each other.

# Elliptical profiles

There are three independent parameters for generating the profiles.A lot combinations of these parameters were studied in detail. For a given  $X_m$  and H, the variation of  $U_2$  and  $S_m$  are insignificant with change in T. Hence one can choose T equal to -1.5. Increasing the values of both  $X_m$  and H improved  $S_m$  but increases  $U_2$  also. From the considerations of reasonable non-uniformity and sufficient separation margin it was found that range of H between 0.4 and 0.5 and X<sub>m</sub> from 1.0 to 1.2 need only be considered. From these analysis, for H=0.4 and H=0.5(T=-1.5) cross plots of U<sub>2</sub> with  $S_m$  as  $X_m$  varies from 0.8 to 1.4 were obtained and they are also shown in Fig. 3.

#### **Curvature Defined Profiles**

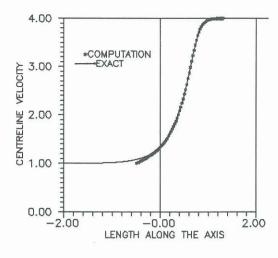
Studies carried out for various combinations of n1 and n2 indicated that n1=1.0 and n2=3.0 was the optimum choice from the considerations of preventing boundary layer separation on the wall and minimising the non-uniformity at the exit. Here also analysis was carried out for various values of  $X_m$ , H for varying T, and  $U_2$  and  $S_m$  were found out. The variation of  $U_2$  with  $S_m$  for combinations of  $X_m$  and H are shown in Fig .2. The envelope of these curves which gives minimum  $U_2$  for a given  $S_m$  is also shown in Fig 2. This envelope is also plotted in Fig. 3 to permit comparison of these profiles.

#### COMPARISON OF PERFORMANCE OF VARIOUS PROFILES

From Fig.3, the comparitive performance of various profiles can be assessed.It is obvious that "Curvature Defined Profiles" are far superior to all the other profiles since U<sub>2</sub> is far lower for a given S<sub>m</sub>.For S<sub>m</sub>=0.05, sine, cubic and fifth order polynomial have more or less similar performance, ie, U2 of 1.4%. The elliptical profile with H=0.4 seems to have better performance than cubic and sine profiles but it is seen from Fig.4 that wall velocity variation is not very smooth for the elliptical profile. In any case the performance is nowhere near that of the newprofile for which the wall velocity variation is very smooth. Fig 4 provides comparison of wall and centreline velocitiy distribution for all the profiles analysed for a separation margin of 0.05.

### CONCLUSIONS

For the first time a systematic comparison of the performance of 2-D contraction profiles has been made. Elliptical profiles do not seem to have any advantage over the sine and cubic profiles except at very high margin of separations with corresponding high values of flow non-uniformity at the exit. Sine and cubic profile performances are more or less same. The Curvature Defined Profiles offer very low non-uniformity compared to other profiles for the same margin of separation.



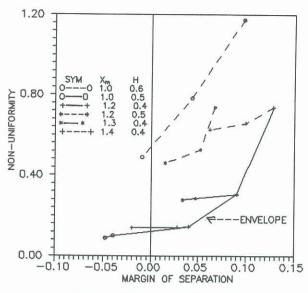


FIG 2. CURVATURE DEFINED PROFILES TO ARRIVE AT MINIMUM  $\rm U_2$  FOR GIVEN  $\rm S_m$ 

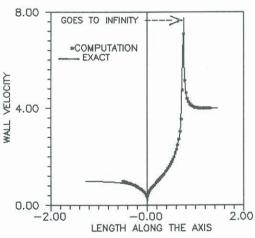




FIG 1 COMPARISON OF WALL AND CENTRELINE VELOCITIES FOR 2-D DUCT WITH STRAIGHT WALLS

WALL

CENTRELINE

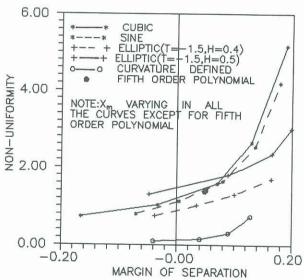


FIG 3. COMPARISON OF PERFORMANCE OF VARIOUS CONTRACTION PROFILES

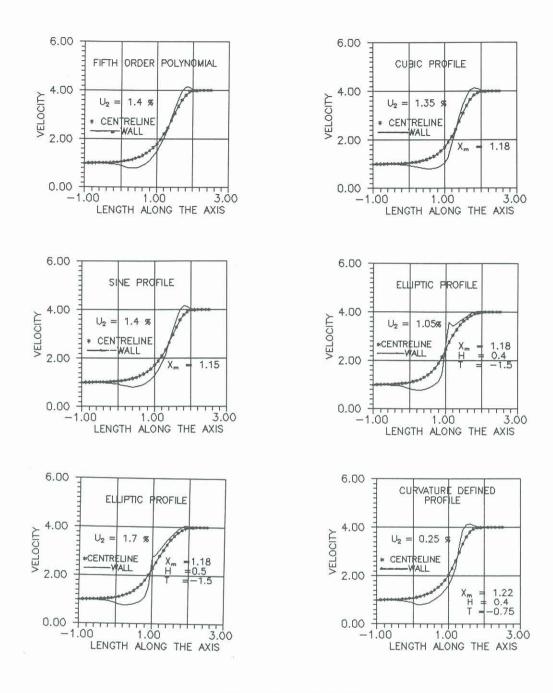


FIG 4 WALL AND CENTRELINE VELOCITY DISTRIBUTION FOR ALL THE CONTRACTION PROFILES FOR A SEPARATON MARGIN OF 0.05

# REFERENCES

- 1. Downie J.H et al,1984,"On the design of 3-D wind tunnel contractions", Aeronautical Journal
- 2. Mehta R. D. and Bradshaw P., 1979, "Design rules for small low speed wind tunnels", Aeronautical Journal, pp443-449
- 3. Morel T. 1975, "Comprehensive design axisymmetric wind tunnel contractions", Jou. of Fluids Engg, pp225-233.
- 4.Stratford B.S.1959,"The prediction of separation of the turbulent boundary layer, Jou. Fluid Mechanics, 5, 1.